

Quantum satellites and tests of relativity



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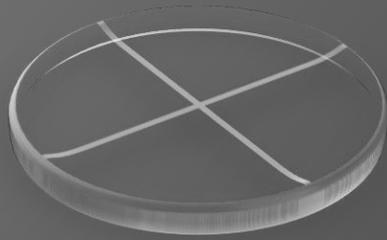
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*PHOTONS AS A GRAVITY
PROBE*



**International Workshop on Relativistic
Quantum Information (RQI-N)**

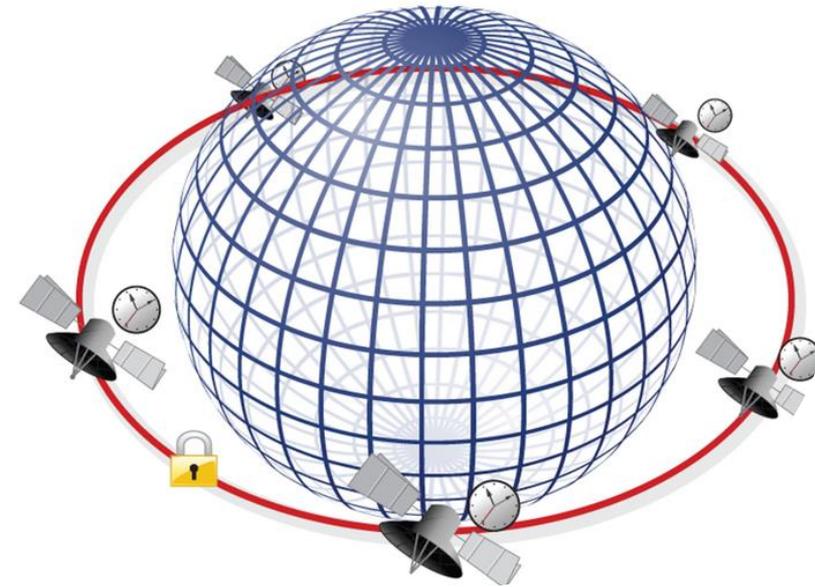
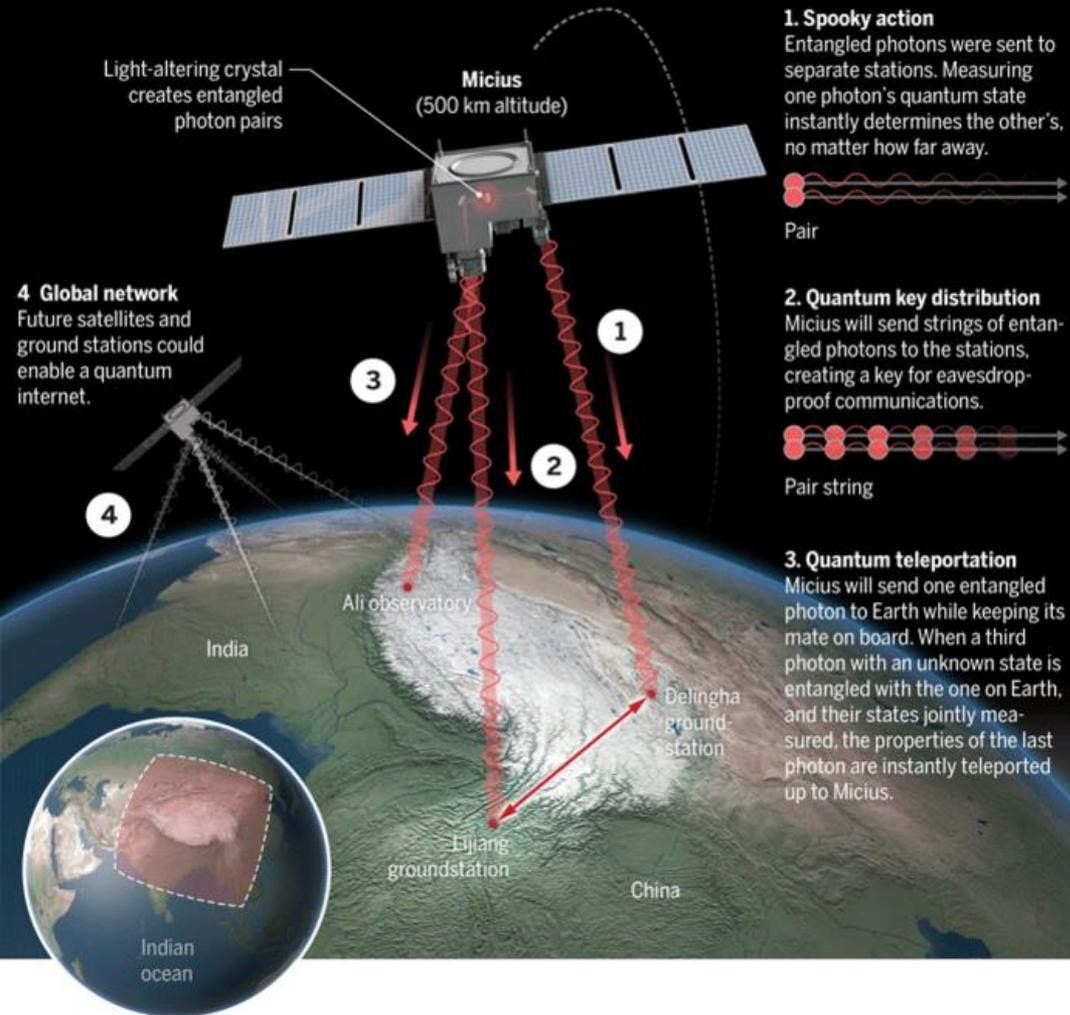
May 28-30, 2010

Lecture Hall I, Science and Engineering Building, NDHU

Raison d'être: sat-based quantum technologies

Quantum leaps

China's Micius satellite, launched in August 2016, has now validated across a record 1200 kilometers the "spooky action" that Albert Einstein abhorred (1). The team is planning other quantum tricks (2-4).



❖ Space-based entangled optical clock array

✓ Space-based QKD

Outline

- $EEP = WEP_{1,2} + LLI + LPI_{1,2}$
- Interferometry & PPN
- Sat LPI_1 experiment
- Spin & WEP [+rotation]

DRT, F. Vedovato, M. Schiavon, A. R. H. Smith, P. Magnani,
G. Vallone, and P. Villoresi

Proposal for an optical test of the Einstein Equivalence Principle,
arXiv: 1811.04835 (2018)

S.Ghosh, L.-C.Kwek, DRT, S. Vinjanampathy,
Detecting beyond standard model physics via weak-value magnetometry,
soon



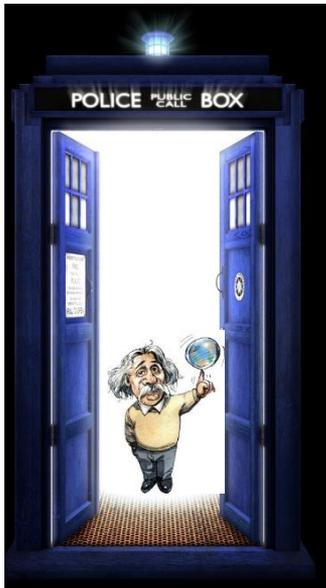
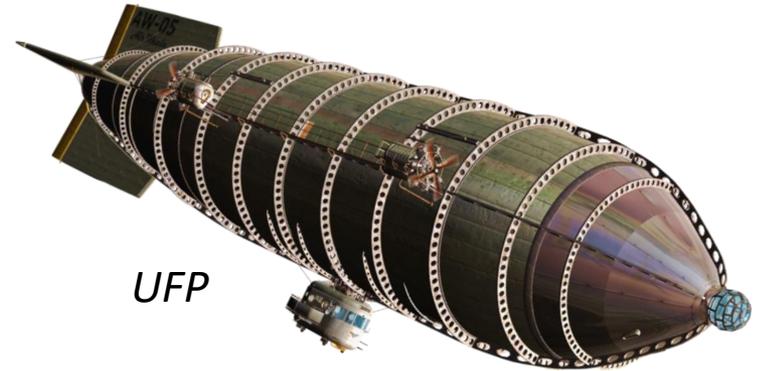
Einstein Equivalence Principle

Weak Equivalence Principle

The trajectory of a freely falling test body is independent of its internal composition.

SM
invariants

$$m_{\text{inertial}} = m_{\text{passive grav}}$$



=



Einstein's elevator:

IF all bodies fall with the same acceleration in an external gravitational field, then to an observer in a small freely falling lab in the same gravitational field, they appear unaccelerated.

Local Invariances

Local Lorentz Invariance:

outcomes of experiments are independent of the velocity of the laboratory where the experiment takes place.

Local Position Invariance:

the outcome of any local non-gravitational experiment is independent of (1) where and (2) when in the universe it is performed

where:

gravitational red shift

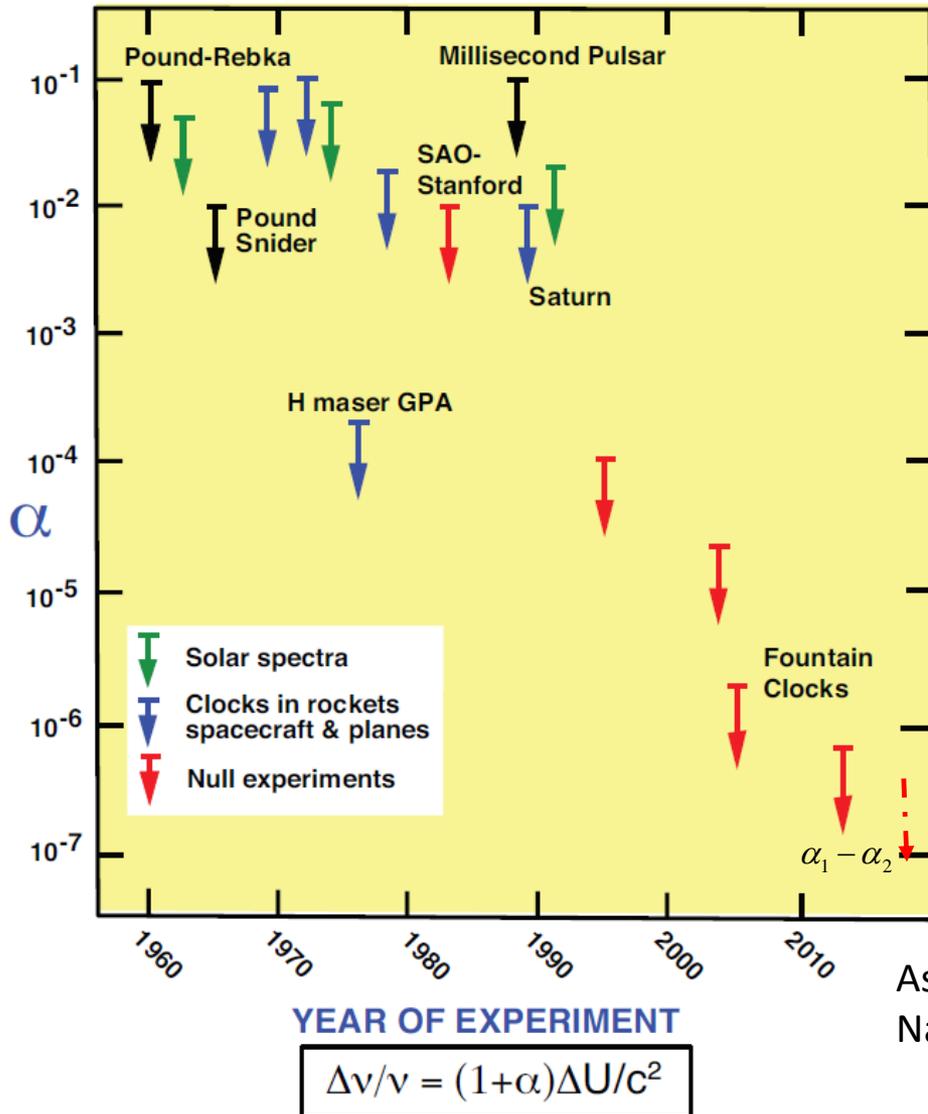
when:

variability of physical constants

$$EEP = WEP + LLI + LPI_{1,2}$$

Bounds: what do we know about the redshift?

Will, Liv. Rev. Rel. **17**, 4 (2014)



Parameterization of the equivalence principle violation

$$g_{00} = -1 + (1 + \alpha)U/c^2 + \dots$$

To be noted:

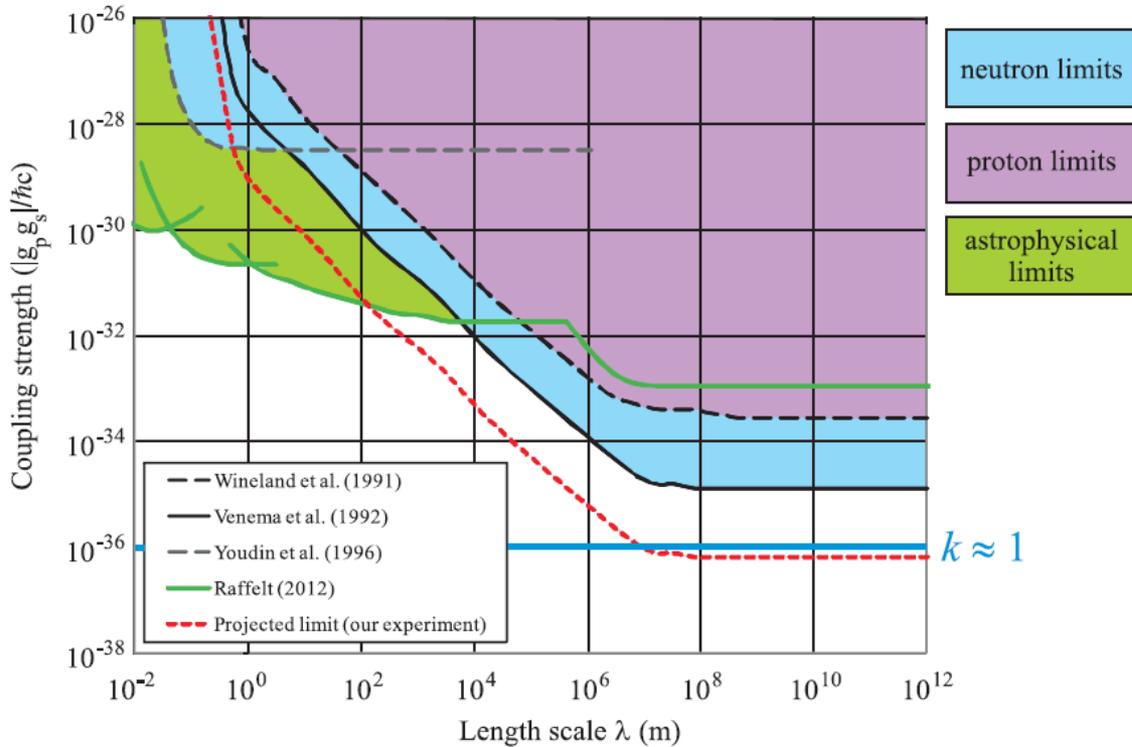
- the most precise tests compare fermions [EM is a messenger]
- <many> base the frequency standard on fermions

A wish:

In SME different sectors (may) couple differently, so it would be nice to have a single-source all-optical test

Ashby, Parker, Palta,
Nature Phys **14**, 802 (2018)

Bounds: what do we know about spin coupling?



Parameterization of the direct spin gravity coupling

$$H_{\text{ext}} = \frac{\hbar k}{2c} \vec{a} \cdot \vec{\sigma}$$

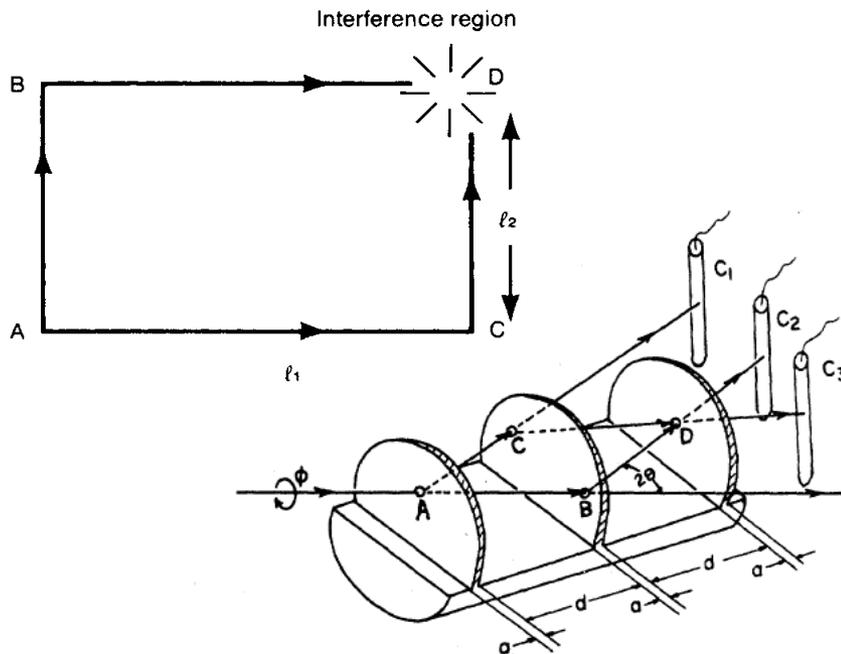
To be noted:

- Rotating frame effects

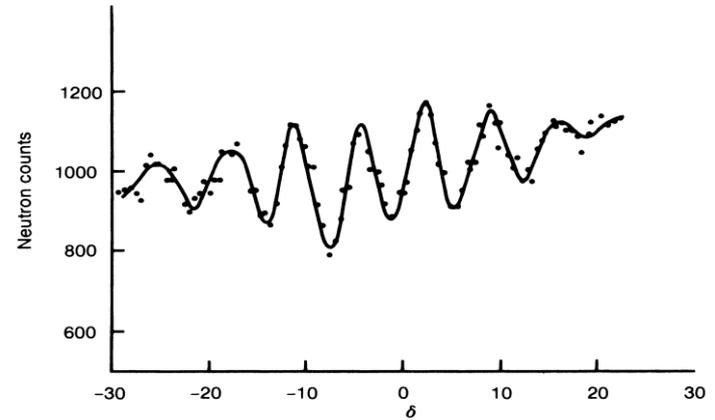
Kimball et al, Ann. Phys. (Berlin) **525**, 514 (2013)

Idea: COW with photons

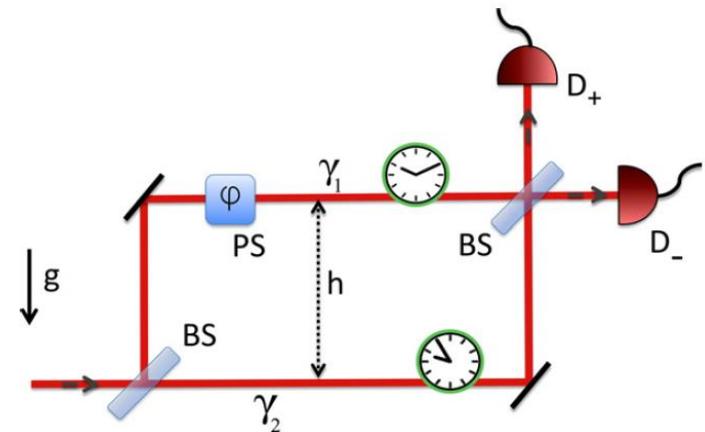
thermal neutrons, $\lambda = 1.42\text{\AA}$



- velocity ~ 2.9 km/sec
- size of the arms 3.1-3.5 cm.
- A, D are beam splitters (silicon slabs)
- B, C are mirrors (actually also beam splitters)



Colella, Overhauser, Werner 1975



Zych, Costa, Piovski, Brukner, Nature Comm. **2**, 505 (2011)

Idea: COW with photons

Back of the envelope calculation:

- mass-energy equivalence,
- Newtonian photons

$$\Delta\psi = \frac{2\pi}{\lambda} \frac{ghq}{c^2}$$

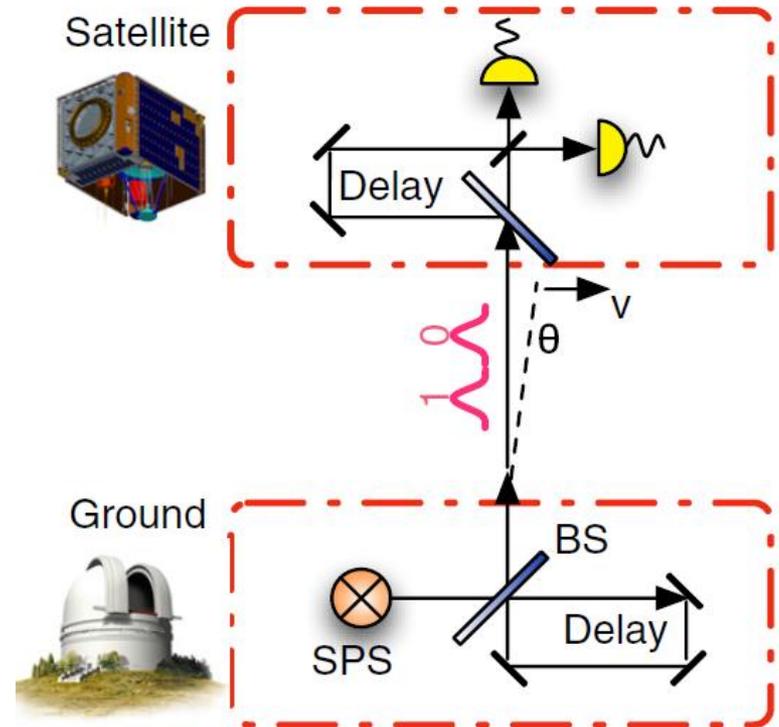
$$h \sim 300 \text{ km}$$

$$q = 6 \text{ km}$$

$$\lambda = 800 \text{ nm}$$

$$\Delta\psi \sim 2 \text{ rad}$$

Potential problems: factors of 2



Rideout *et al.*,
Class. Quant. Grav. **29**, 224011 (2012).



Interferometry

Geometric optics

Shortwave asymptotics

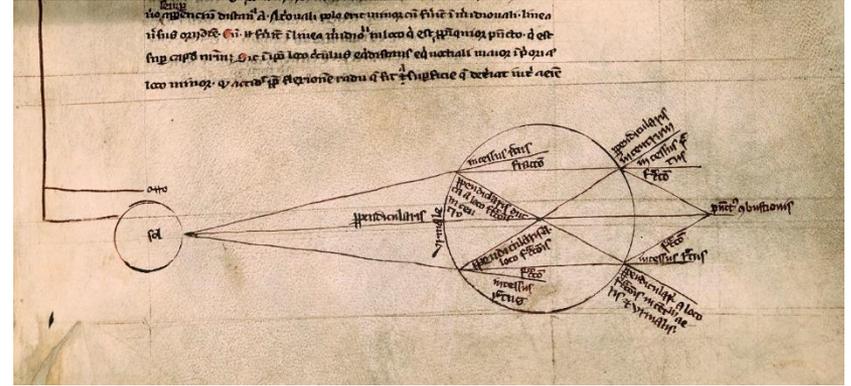
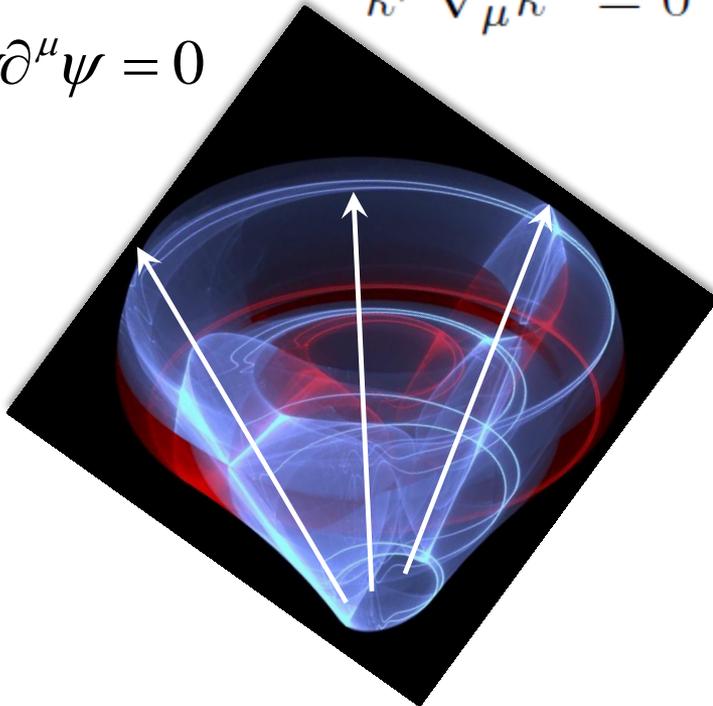
$$A^\mu = a^\mu e^{i\psi}$$

Substitute, gauge-fix, expand

$$k_\mu = \partial_\mu \psi$$

$$k^\mu \nabla_\mu k^\nu = 0$$

$$\partial_\mu \psi \partial^\mu \psi = 0$$



Leading order:
eikonal equation

Trajectories: photons as massless point particles that move on the rays prescribed by geometric optics.

Hamilton-Jacobi

stationary spacetimes

Metric is time-independent.
Exists a timelike Killing vector ξ

Meaning:
Conservation of energy

Conserved frequency

$$k^\mu \xi_\mu = -\omega_\infty / c = \text{const} \quad \rightarrow -k_0$$

Local frequency

Reference frame moves with u_F^μ $\omega_F = -k \cdot u_F$

Phase

HJ equation separates, so the phase takes the usual form

$$\psi(t, \vec{x}) = -\omega_\infty(t - t_0) + \omega_\infty S(\vec{x}, \vec{x}_0) \quad \blacktriangleleft \quad \partial_\mu \psi \partial^\mu \psi = 0$$

Physical optics

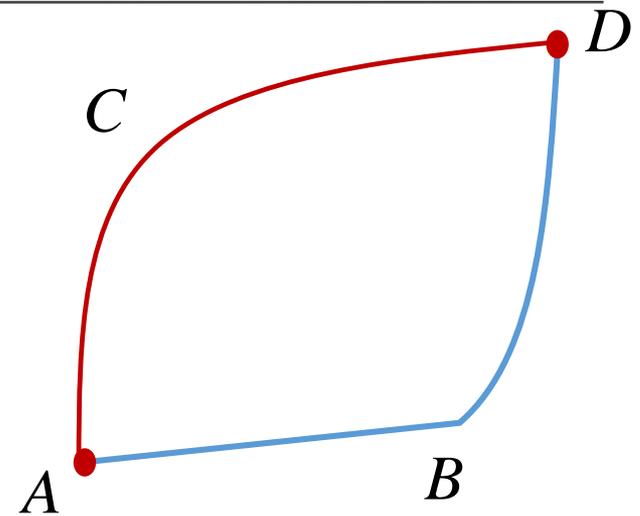
$$I_{12}(t_D, \vec{x}_D) \propto \cos \Delta\psi(t_D, \vec{x}_D)$$

$$\Delta\psi(t_D, \vec{x}_D) := \psi_{ABD}(t_D, \vec{x}_D) - \psi_{ACD}(t_D, \vec{x}_D)$$

Classical modes

Quantum detection/interpretation

$$a \rightarrow \hat{a}$$



Phase difference, stationary spacetime

$$\Delta\psi(t, \vec{x}_D) = \omega_\infty S_{ABC}(\vec{x}_D, \vec{x}_A) - \omega_\infty S_{ACD}(\vec{x}_D, \vec{x}_A)$$

$$= \omega_\infty (t_{ABD} - t_{ACD}) = \omega_\infty \Delta t$$

Two geodesics ▲



Phase difference

Segments, boundary conditions & Lorentz

$$\psi(t, x) = -\omega_n(t - t_{n-1}) + \omega_n \Delta t(\vec{x}; \vec{x}_{n-1}) + \sum_{k=1}^n \phi_{k-1}$$



Lorentz transforms

The phase a scalar

$$\psi(x) = k^a x_a + \Phi_1$$

$$\psi(x) = \psi'(x') = k^{a'} x_{a'} + \Phi_1$$



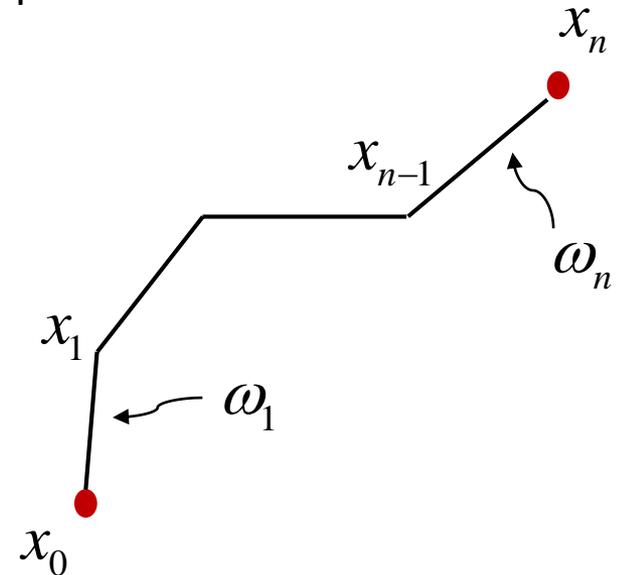
Mirrors: boundary conditions, Doppler



Delay

(approximately) @ one point \vec{x}_F
for a proper time τ_F

$$\phi_F = \omega_F \tau_F$$





PPN

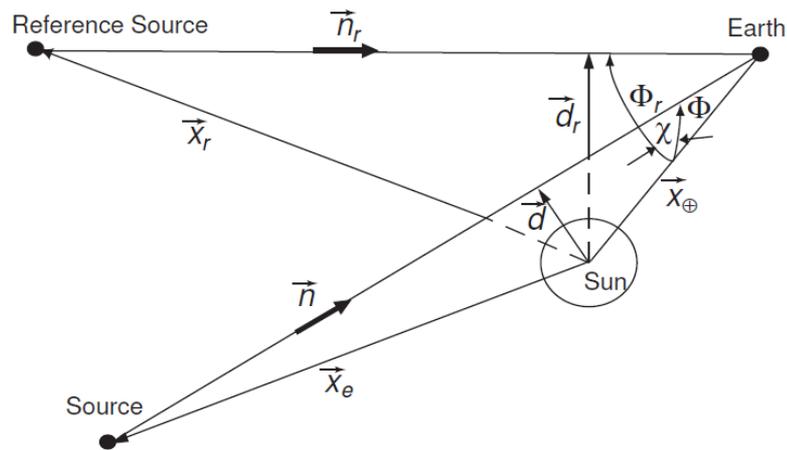
Parameterized post-Newtonian

a systematic method for studying a system of slowly moving bodies bound together by weak gravitational forces

Newtonian limit:

$$g_{00} = -(1 + 2\varphi/c^2)$$

$$g_{ij} = \delta_{ij}$$



$$U := -\varphi/c^2$$

EoM for test particles including the post-Newtonian effects

Making sense of what to keep:

$$\frac{GM}{rc^2} \sim \frac{v^2}{c^2} \equiv \beta^2 \sim \varepsilon^2$$

PPN

for photons

Frame:

Earth-centered, inertial

Earth:

Rigid, uniformly rotating
[spherical/elliptical]

Metric:

$$ds^2 = -V^2(r)c^2 dt^2 + \vec{R} \cdot d\vec{x} c dt + W^2(r) d\vec{x} \cdot d\vec{x}$$

$$V(r) = 1 - \epsilon^2 \frac{U}{c^2}, \quad W(r) = 1 + \epsilon^2 \gamma \frac{U}{c^2}$$

$$\vec{R} = -\epsilon^3 2(1 + \gamma) \frac{G}{c^3} \frac{\vec{J} \times \vec{x}}{r^3}$$

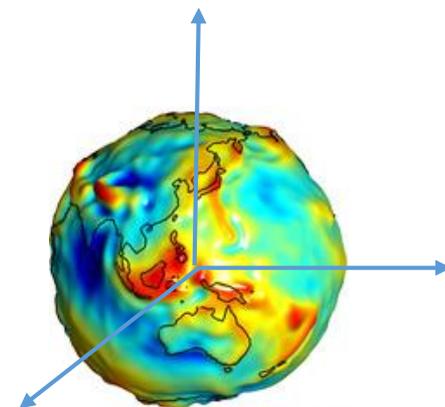
▲ Lense-Thirring aka frame dragging ▲

Scale:

$$\frac{GM}{rc^2} = \frac{v^2}{c^2} \sim \epsilon^2$$
$$\epsilon = 2.6 \times 10^{-5}$$

$$U = \frac{GM}{r} Q \simeq \frac{GM}{r}$$

▲ Newton & co ▲



The Earth
is not round

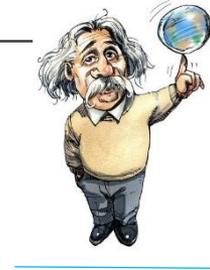


$$Q = 1 - \frac{1}{2} J_2 \frac{R^2}{r^2} (3 \cos^2 \theta - 1)$$

$$J_2 = 1.08 \times 10^{-3}$$

PPN

for photons



$$\vec{x}(t) =: \vec{x}_{(0)}(t) + \epsilon^2 \vec{x}_{(2)}(t)$$

EoM

$$\frac{d^2 \vec{x}_{(2)}^\perp}{dt^2} = (1 + \gamma) (\vec{\nabla} U - \hat{n}(\hat{n} \cdot \vec{\nabla} U))$$

$$\frac{dx_{(2)}^{\parallel}}{dt} = -(1 + \gamma) \frac{U}{c},$$

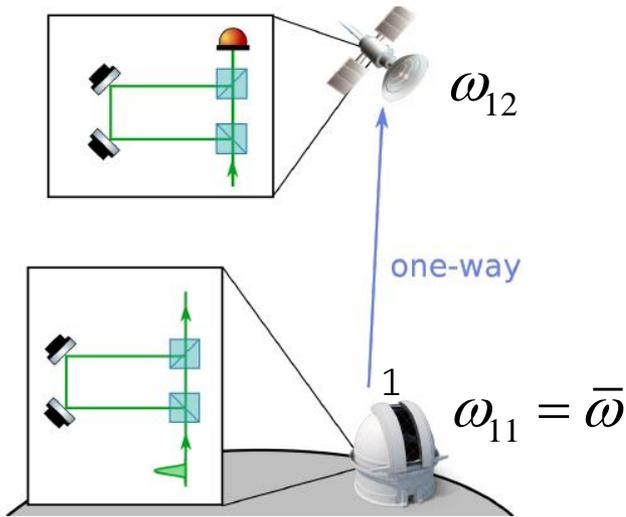
A useful consequence of massless particles

Time delay

$$c\Delta t = |\vec{x} - \vec{b}| + (1 + \gamma) \frac{GM}{c^2} \ln \frac{r + \vec{x} \cdot \hat{n}}{b + \vec{b} \cdot \hat{n}}$$

Phase difference

PPN +SR

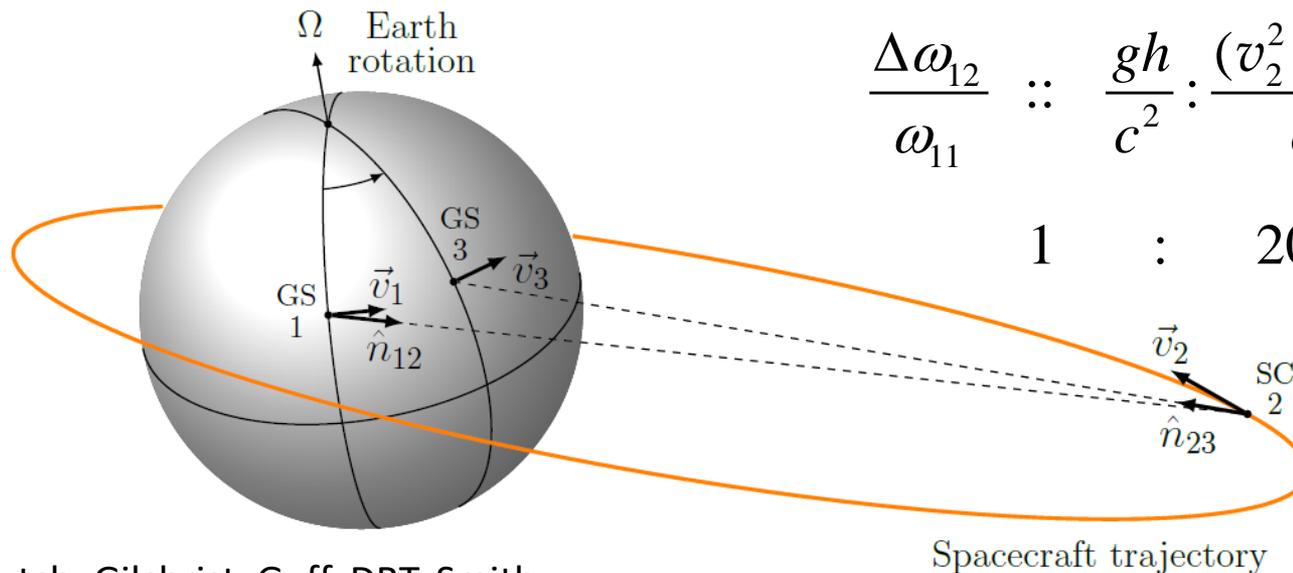


$$\Delta\phi = (\omega_{12} - \omega_{11})\tau \quad \text{Condition: } \Delta t \ll \tau_c$$

Frequency shift for the pulse @ sat

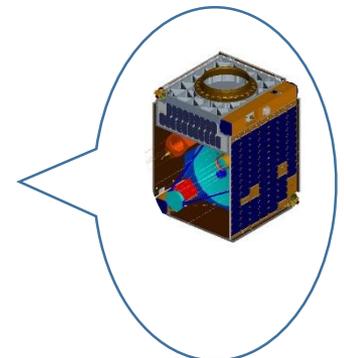
$$\frac{\omega_{12}}{\bar{\omega}} = \left(\frac{1 - U_E - \frac{1}{2}\beta_E^2}{1 - U_2 - \frac{1}{2}\beta_2^2} \right) \left(\frac{1 - \hat{n}_{12} \cdot \vec{\beta}_2}{1 - \hat{n}_{12} \cdot \vec{\beta}_1} \right) + \text{👤}$$

The Doppler problem of the optical COW

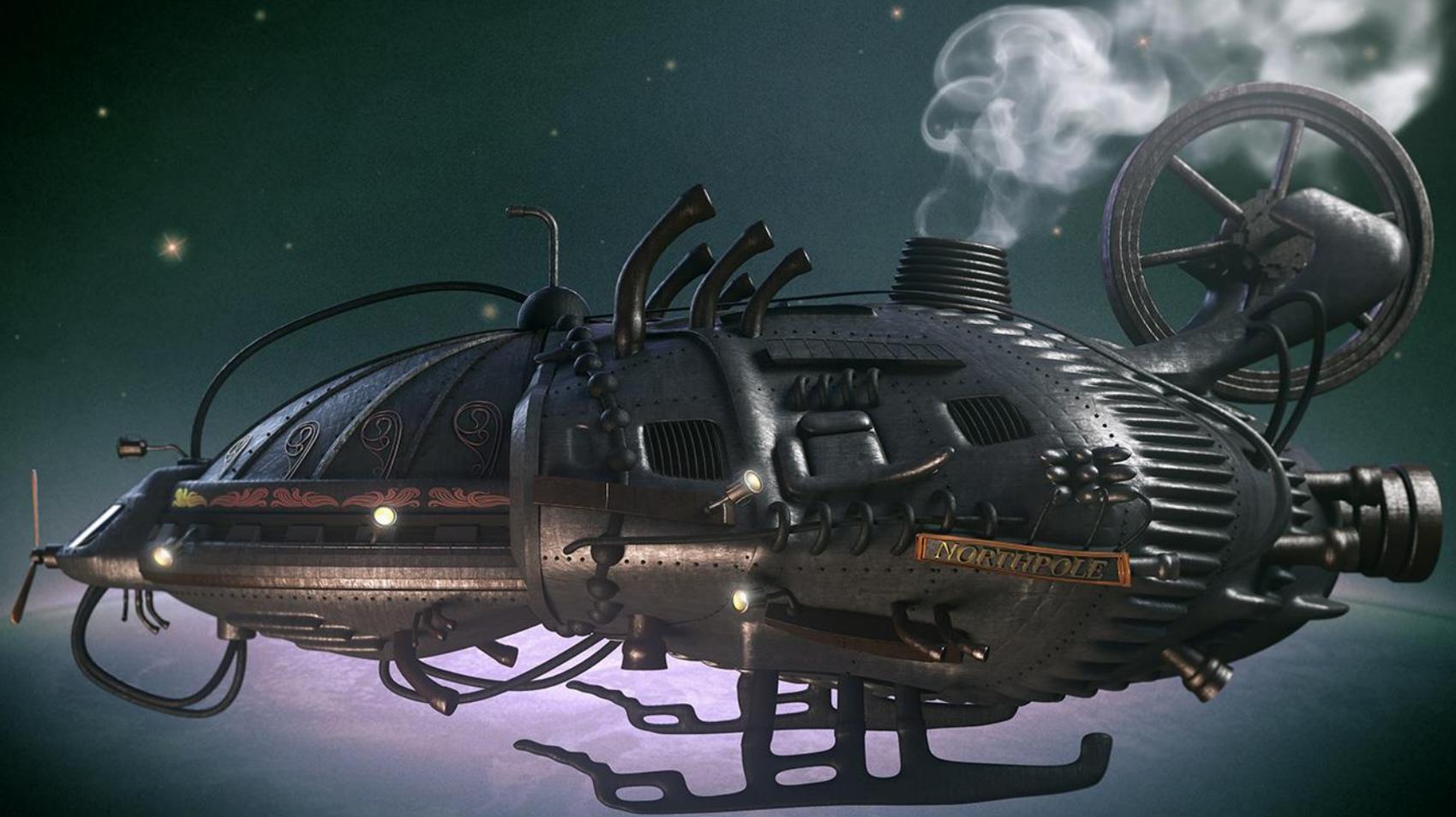


$$\frac{\Delta\omega_{12}}{\omega_{11}} \quad \text{::} \quad \frac{gh}{c^2} \quad \text{:} \quad \frac{(v_2^2 - v_1^2)}{c^2} \quad \text{:} \quad \frac{(\vec{v}_2 - \vec{v}_1) \cdot \hat{n}_{12}}{c}$$

$$1 \quad \text{:} \quad 20 \quad \text{:} \quad 10^5$$

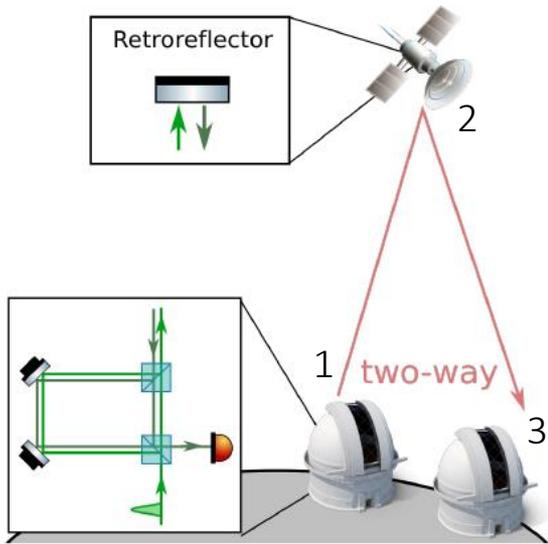


Spacecraft trajectory



Sat LPI_1 experiment

Doppler and qubits



Scheme of the experiment and satellite radial velocity

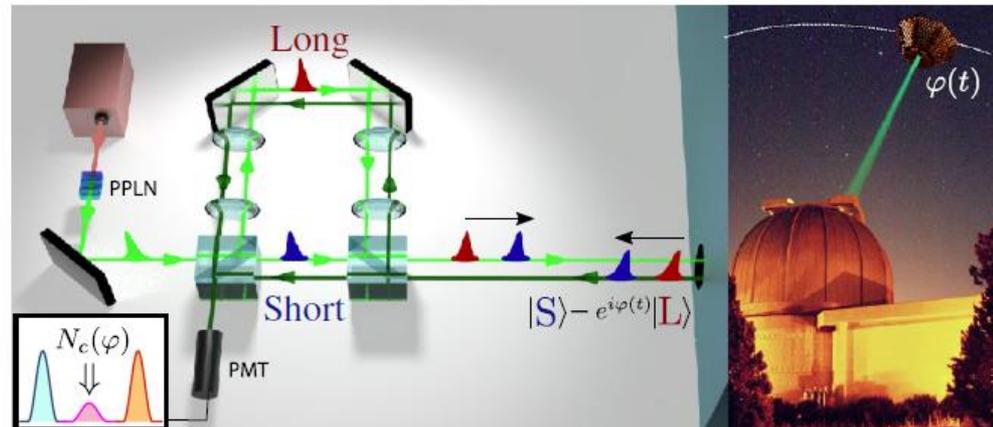
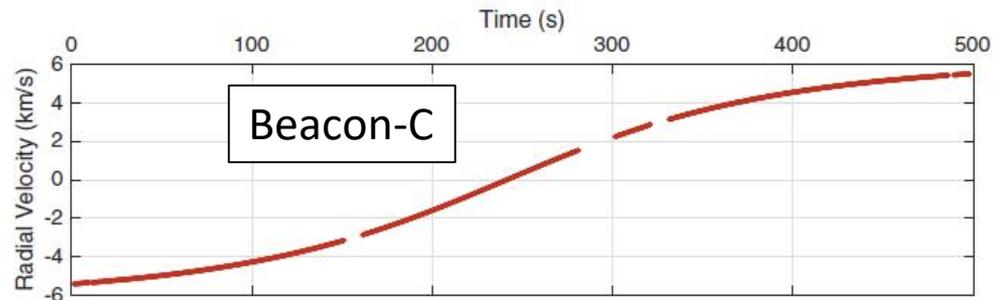
Bottom: the unbalanced MZI with the two 4f systems used for the generation of the state and the measurement of the interference. The light and dark green lines represent the beams outgoing to and ingoing from the telescope.

Inset: the [expected] detection pattern

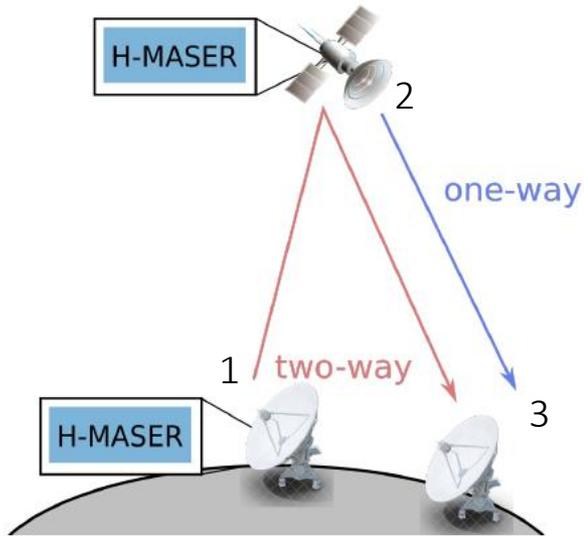
$$\frac{\omega_{13}}{\bar{\omega}} = \left(\frac{1 - \hat{n}_{23} \cdot \vec{\beta}_3}{1 - \hat{n}_{23} \cdot \vec{\beta}_2} \right) \left(\frac{1 - \hat{n}_{12} \cdot \vec{\beta}_2}{1 - \hat{n}_{12} \cdot \vec{\beta}_1} \right) + \text{[Cartoon Character]}$$

Interference at the single photon level along satellite-ground channels: successful simulation of quantum communication that is based on the 1st order Doppler

Vallone *et al.*, Phys. Rev. Lett. **116**, 253601 (2016)



removing Doppler



GP-A solution

Two sources (1 & 2). Two detections at 3.

The 1st order Doppler is removed in

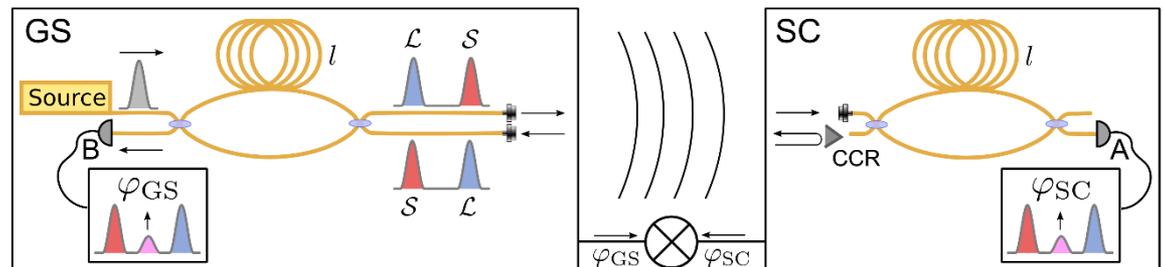
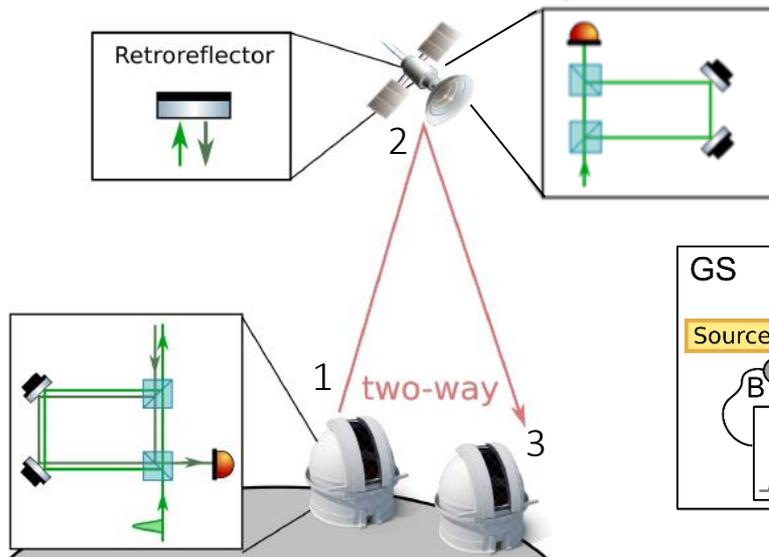
$$\Omega = \omega_{23} - \frac{1}{2} \omega_{13}$$

Vessot and Levine, *NASA technical report NASA-CR-161409* (1979)

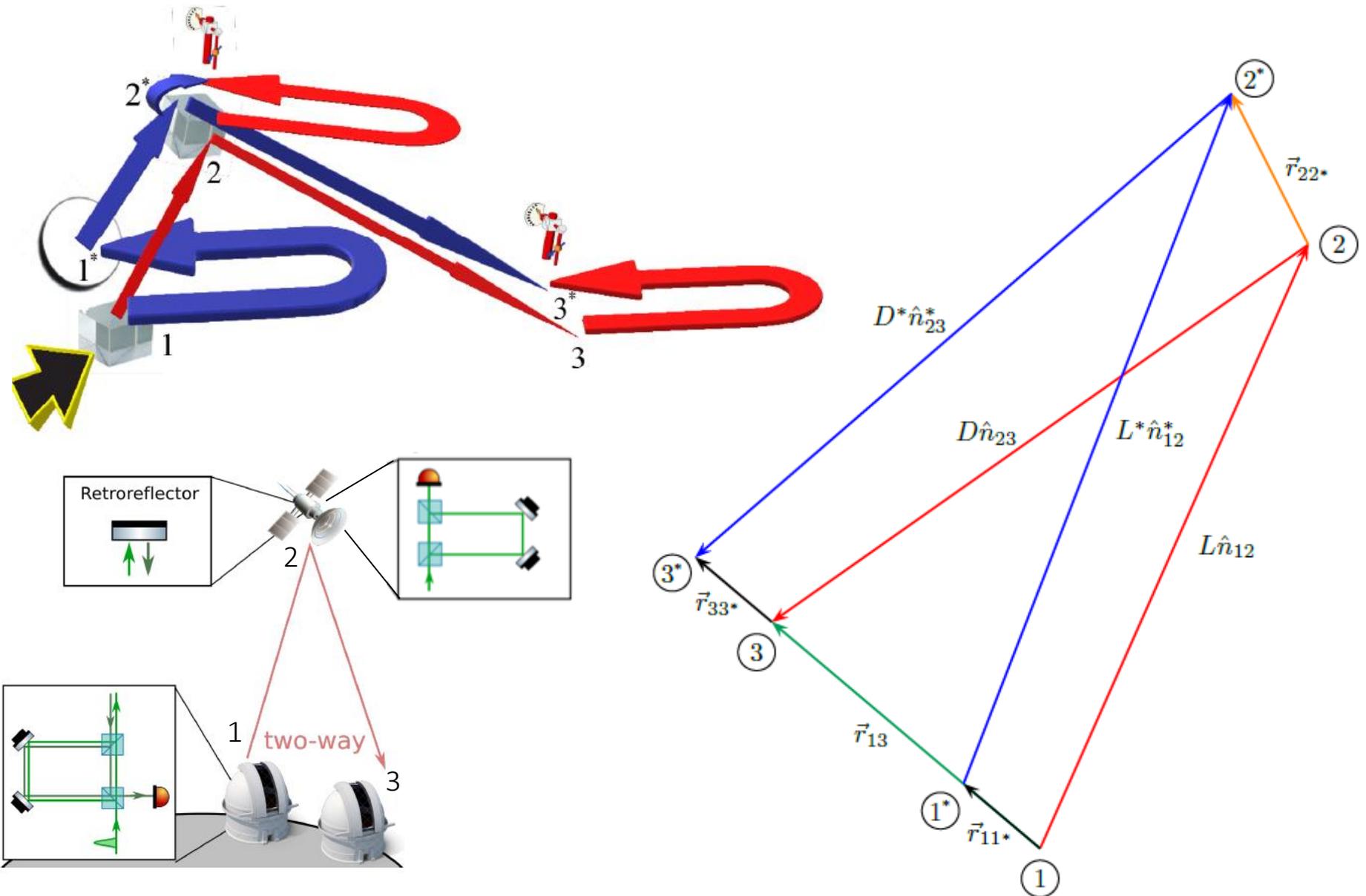
A 1+2 way optical COW

+Time-delay interferometry

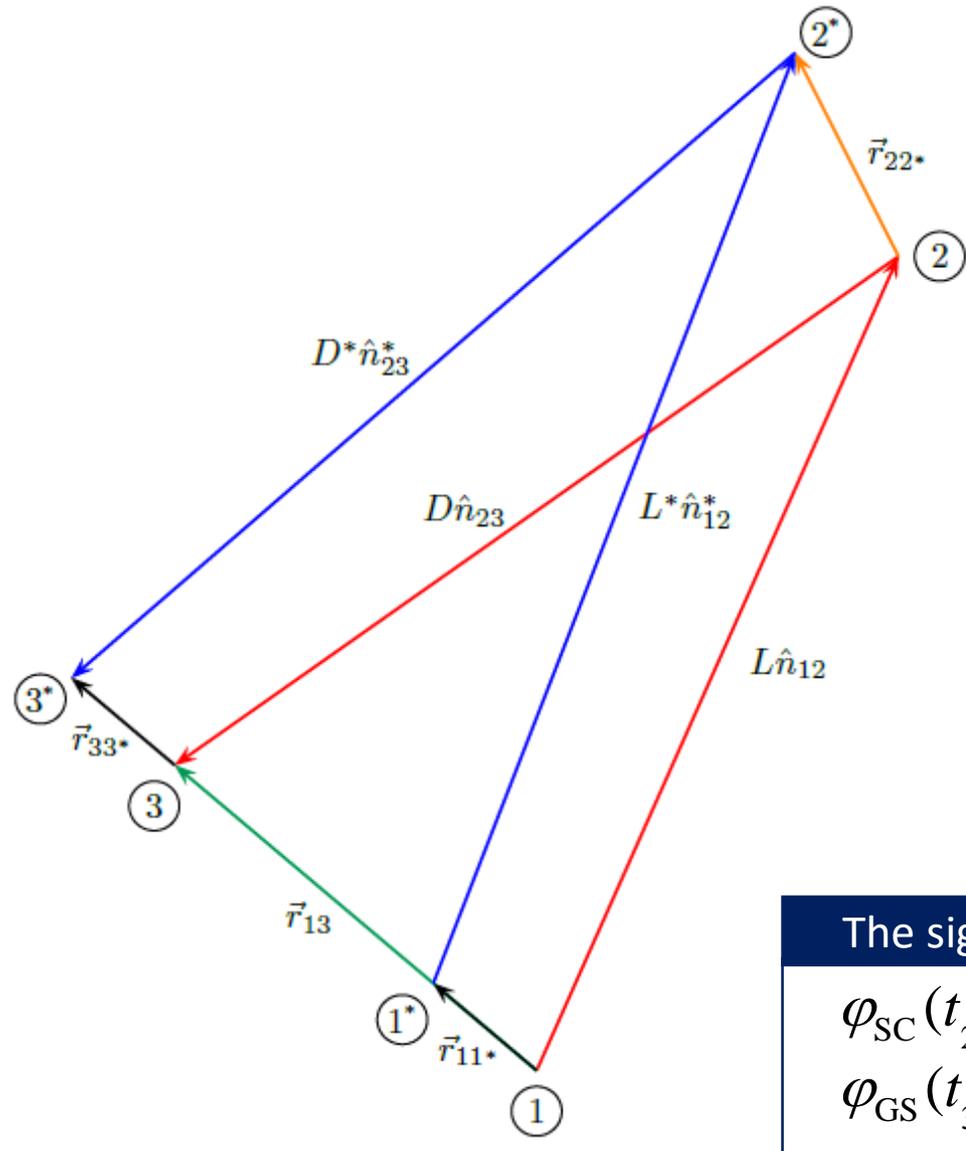
Tinto and Dhurandhar, *Living Rev. Relativity* **17**, 6 (2014).



Geometry



interferometry and PPN



Positions of the ground station and the satellite at different stages of the experiment.

Distances travelled by the beam 1 on the go-return trip are L and D , respectively.

Proper delay times: $\tau = nl/c$

Propagation time (0th order): $T = L/c$

Three useful frames: global, GS, SC

Another expansion parameter:

$$\mu = l/L \sim 10^{-4} \dots 10^{-5}$$

The signals

$$\varphi_{\text{SC}}(t_{2^*})$$

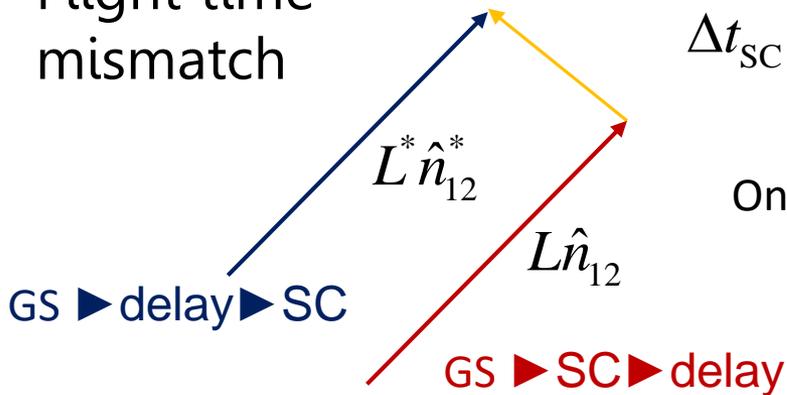
$$\varphi_{\text{GS}}(t_{3^*})$$

$$\varphi_{\text{SC}}(t_{2^*}) \approx (\omega_{12} - \omega_{11})\tau$$

$$\varphi_{\text{GS}}(t_{3^*}) \approx (\omega_{13} - \omega_{11})\tau$$

the signal

Flight time mismatch



$$\Delta t_{\text{SC}} = (L^* - L)/c + \mathcal{O}(\varepsilon^2) = \tau \left((\hat{\mathbf{d}}_2 - \hat{\mathbf{d}}_1) + \mathcal{O}(\varepsilon^2) \right)$$

$$\hat{\mathbf{d}}_i := \hat{n}_{12} \cdot \vec{\mathbf{v}}_i / c$$

On the ground:

$$\Delta t_{\text{GS}} = 2\Delta t_{\text{SC}} + \mathcal{O}(\varepsilon^2)$$

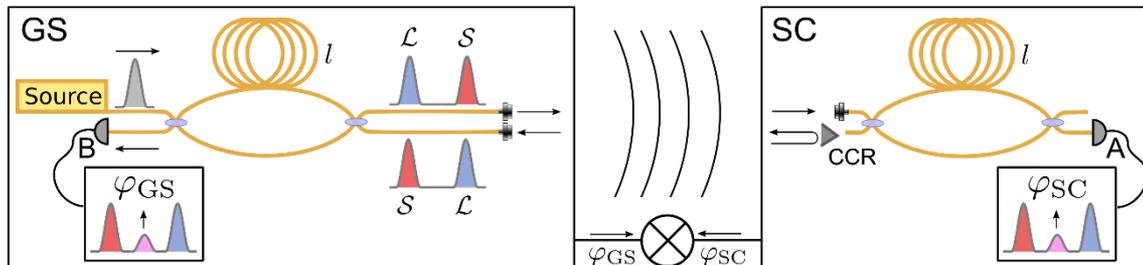
Condition: $\Delta t_{\text{SC}} \ll \tau_c$

Features

- ✓ Have to take into account the delay
- ✓ Subtraction of the signals removes the 1st order Doppler

$$S \left(t_{3^*}^{\text{Earth}}, t_{2^*}^{\text{sat}} \right) = \varphi_{\text{SC}} \left(t_{2^*}^{\text{sat}} \right) - \frac{1}{2} \varphi_{\text{GS}} \left(t_{3^*}^{\text{Earth}}, t_{2^*}^{\text{sat}} \right)$$

Some real-life issues:



- + free-space to single-mode fiber coupling
- + Turbulence
- + delay vs coherence times
- + Different imbalances

the signal

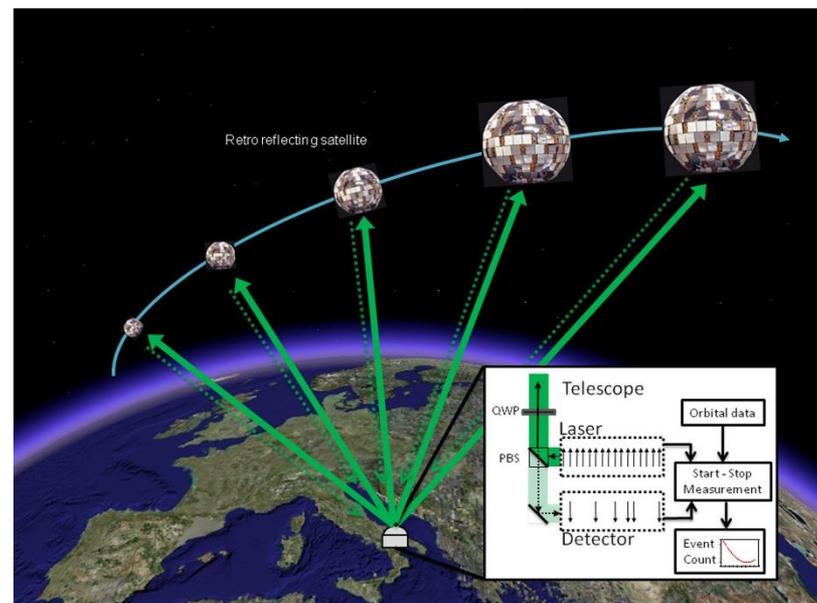
$$\frac{S}{\omega_{11}\tau} = (U_2 - U_1) + \frac{1}{2}(\vec{\beta}_1 - \vec{\beta}_2)^2 - (\mathbf{d}_1 - \mathbf{d}_2)^2 - T\hat{n}_{12} \cdot \vec{a}_1/c$$

$\times(1 + \alpha)$

Looks [very much] like GP-A

Technically difficult, but possible

Simulations for the orbits of some satellites that are observable at Matera RLA

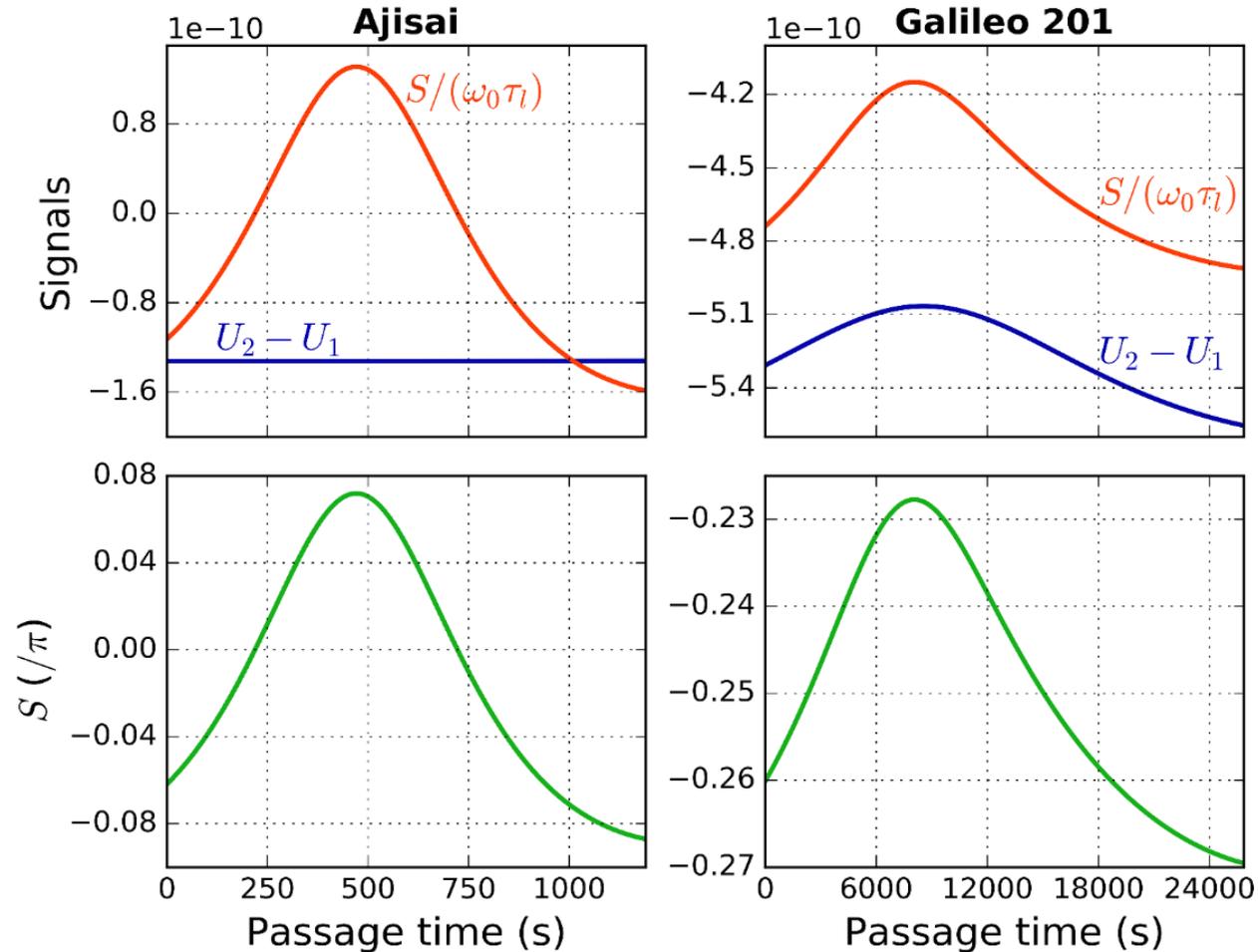


the simulation

as would be seen at MLRA

Ajisai: inclination 50° , eccentricity 0.001, altitude 1,490 km

Galileo 201: inclination 50° , eccentricity 0.158, altitude ranging from 17,000 to 26,210 km





Spin & WEP [+rotation]

(Non-relativistic) spin terms

Mundane & exotic

Spin in a non-inertial frame (linear acceleration and rotation)

Start with Dirac equation on curved background

Do [a/the] FW transform

$$(i\hbar\gamma^\alpha D_\alpha - mc)\psi = 0,$$

(Pick the “large” part; drop the rest mass)

$$H = H_{\text{cl}} + H_{\text{rel}} + H_\sigma + H_{\text{ext}}$$

$$H_{\text{cl}} = \frac{\vec{p}^2}{2m} + m\vec{a} \cdot \vec{x} - \vec{\omega} \cdot L,$$

Hehl and Ni,
Phys. Rev. D **42**, 20145 (1990)

$$H_\sigma = -\frac{1}{2}\hbar\vec{\omega} \cdot \vec{\sigma} + \frac{\hbar}{4mc^2}\vec{\sigma} \cdot (\vec{a} \times \vec{p}).$$

Exotic:

ad-hoc addition, controversial FW transform, or just SME parameters

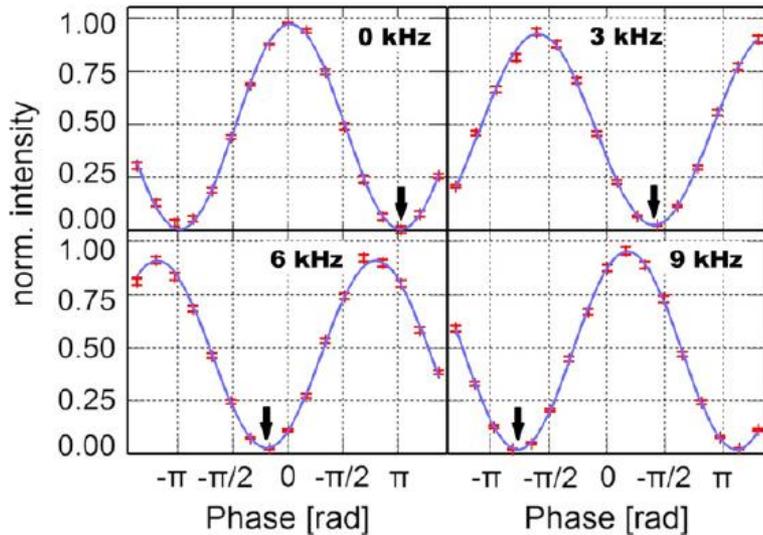
$$H_{\text{ext}} = \frac{\hbar k}{2c}\vec{a} \cdot \vec{\sigma}$$

Peres, Phys. Rev. D **18**, 2739 (1978).

$k = 1$ Obukhov, Phys. Rev. Lett. **86**, 192 (2001).

Status

term	name	observation
$m\vec{a} \cdot \vec{x}$	Bose-Wroblewski	yes
$-\vec{\omega} \cdot L$	Page-Werner	yes
$-\frac{1}{2}\hbar\vec{\omega} \cdot \vec{\sigma}$	Mashhoon	may be
$\frac{\hbar}{4mc^2}\vec{\sigma} \cdot (\vec{a} \times \vec{p})$	Hehl-Ni	no



Polarization of neutrons in rotating magnetic field

Demirel, Sponar, and Hasegawa,
New J. Phys. **17**, 023065 (2015).

Numbers

$$g\hbar/c = 2.15 \times 10^{-23} \text{ eV}$$

$$B_{eq} := \frac{g\hbar}{\mu_B} = 3.72 \times 10^{-19} \text{ Tl}$$

$$\omega_{\text{Earth}} c / g = 2.2 \times 10^3$$

Stability of clocks:

Atomic clocks (microwave) 10^{-16}

Atomic clocks (optical) 10^{-18}

Atomic clocks (quantum) $10^{-17} \dots 10^{-20}$

a hyperfine splitting in ^{133}Cs for $^2S_{1/2}$ is $3.80 \times 10^{-5} \text{ eV}$

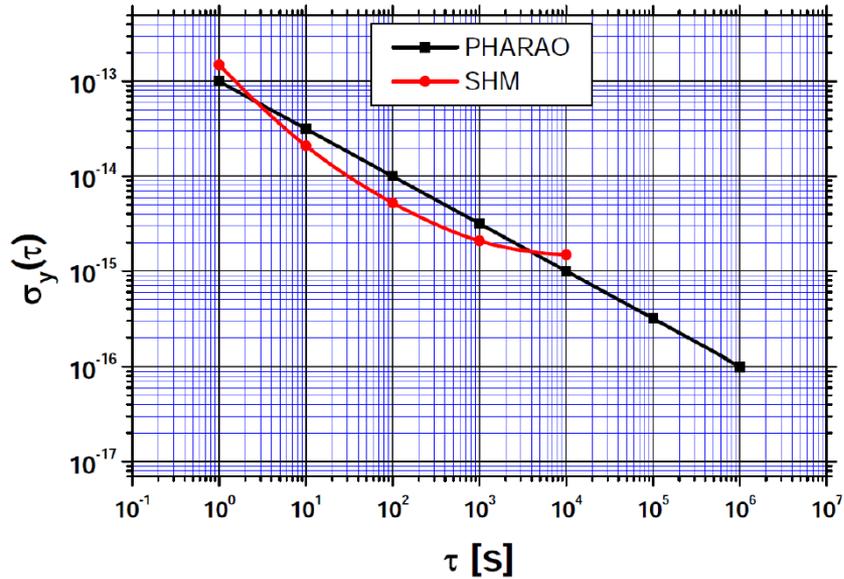
$$\frac{g}{c\Delta\omega_{\text{HF}}} = 3.80 \times 10^{-19}$$



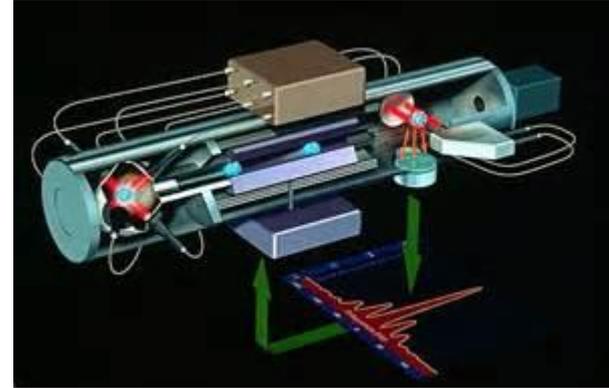
Hinkley *et al.*,
Science **341**, 1215 (2013)

Kómár *et al.*,
Nat. Phys. **10**, 582 (2014)

Atomic clocks



Meynadier et al. *Class. Quant. Grav.* 35, 3 (2018)



ACES goal $10^{-16} \dots 10^{-18}$

Dittus, Lammerzahl, Turischev (eds)
Lasers, Clocks and Drag-Free Control,
(Springer, 2008)

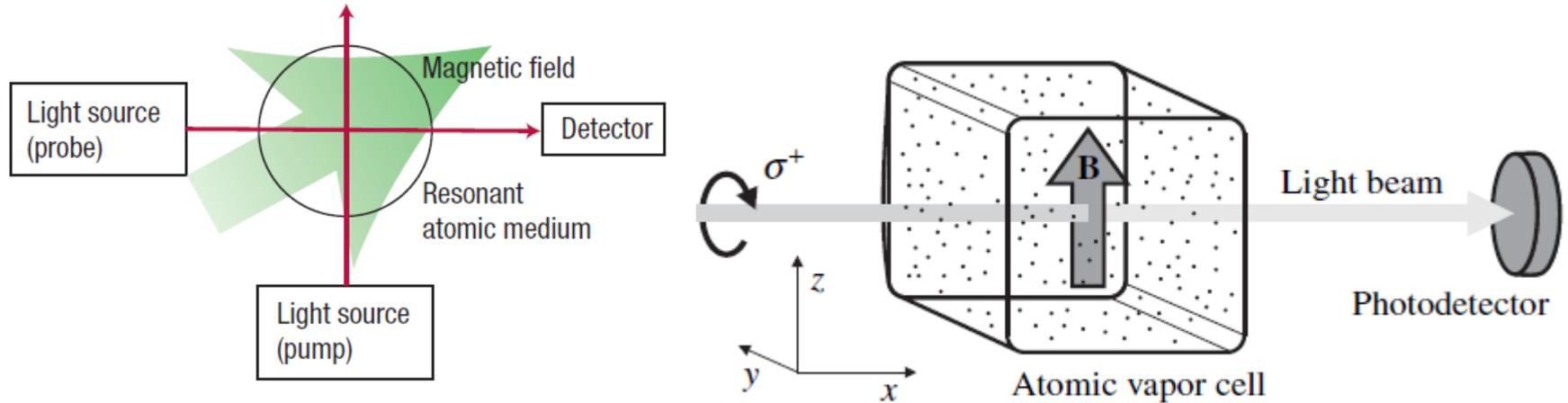
Transitions to be affected

Non-zero ΔM_F

Cs standard: unaffected/insensitive

$$\begin{aligned}
 |F = 4, M_F = 0\rangle &= \frac{1}{\sqrt{2}} \left(|M_S = \frac{1}{2}, M_I = -\frac{1}{2}\rangle + |M_S = -\frac{1}{2}, M_I = \frac{1}{2}\rangle \right) \\
 |F = 3, M_F = 0\rangle &= \frac{1}{\sqrt{2}} \left(|M_S = \frac{1}{2}, M_I = -\frac{1}{2}\rangle - |M_S = -\frac{1}{2}, M_I = \frac{1}{2}\rangle \right)
 \end{aligned}$$

Optical magnetometry



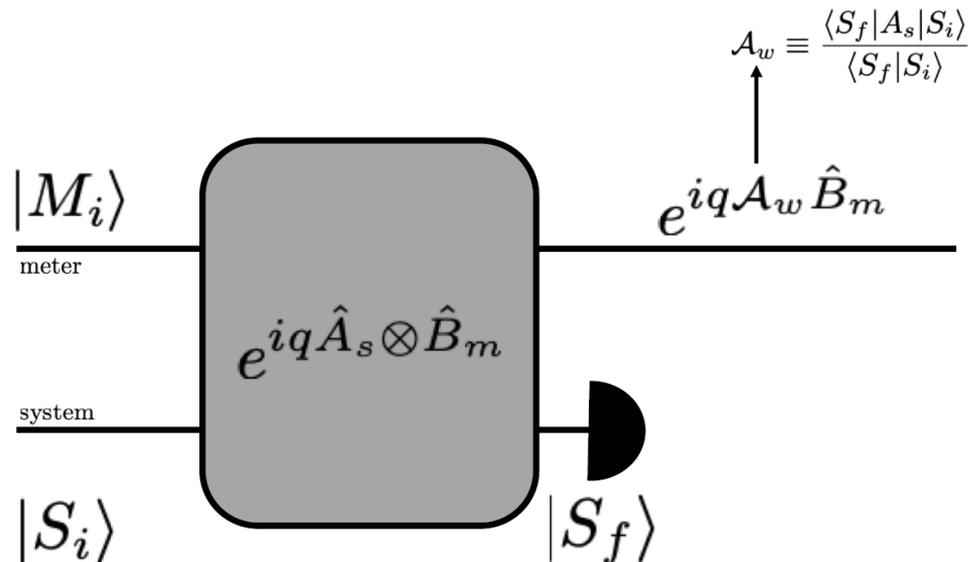
Achieved: $\sim 10^{-15}$ Tl

Planned: 10^{-17} Tl

Budker & Romalis, *Nat Phys.* **3**, 228 (2007)

Budker & Kimbal (eds), *Optical Magnetometry*, (CUP, 2013)

+weak measurements





$$EEP = \text{WEP} + \text{LLI} + \text{LPI}_{1,2}$$

RQI:

how fundamental physics affects quantum info
how quantum info probes fundamental physics